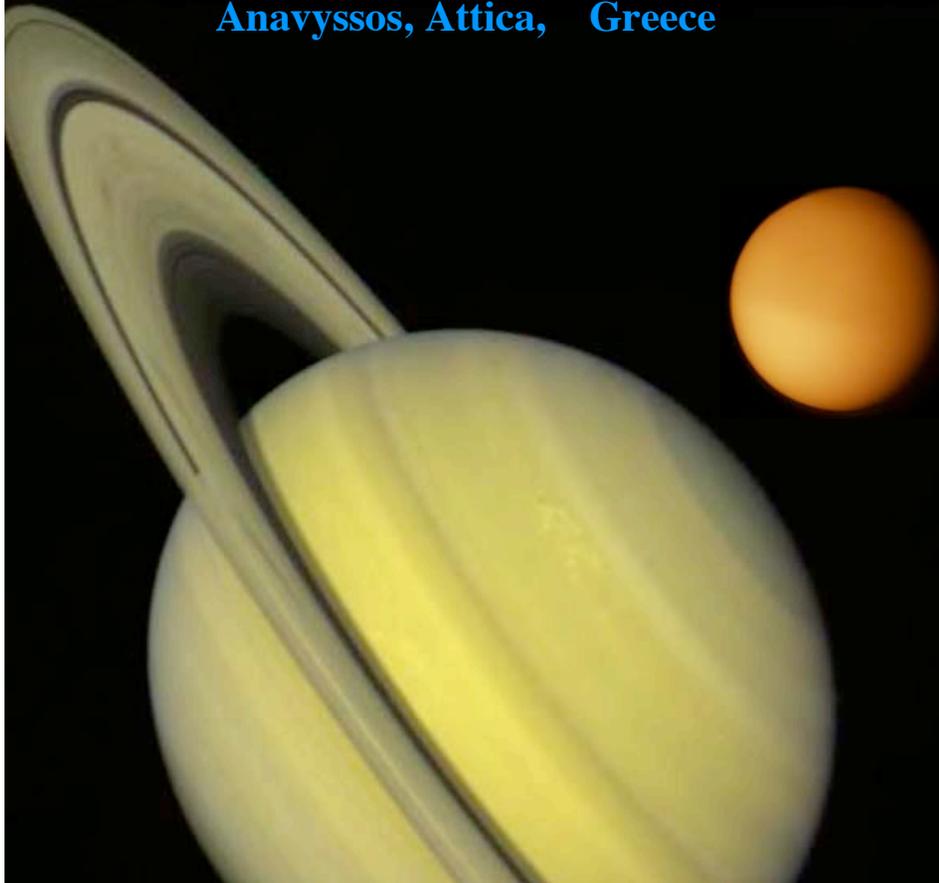


**Current Developments in Future Planetary Probe Sensors:
Heatshield Ablation Detector Adaptation to Mid Density Ablators**

June 27- July 1, 2005

**3rd International Planetary Probe Workshop
Anavyssos, Attica, Greece**



**Ed Martinez, NASA
Tomo Oishi, Eloret
Bill Congdon, ARA
Johnny Fu, Sierra Lobo
Sergey Gorbunov, Eloret**



Outline



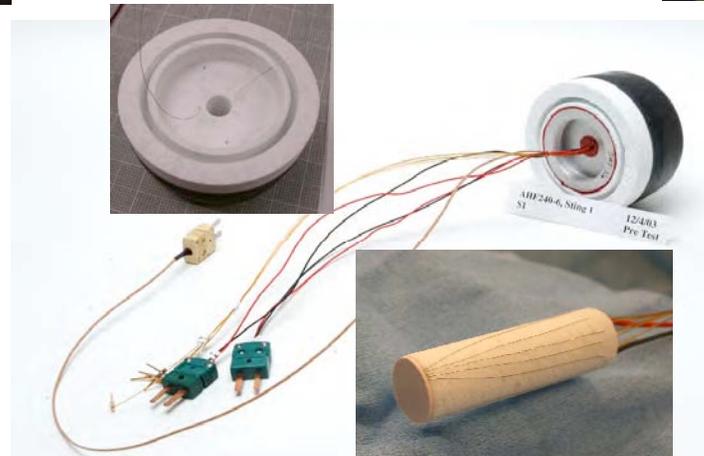
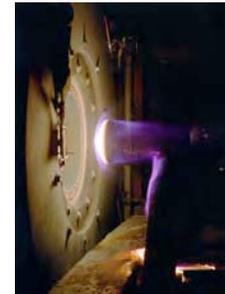
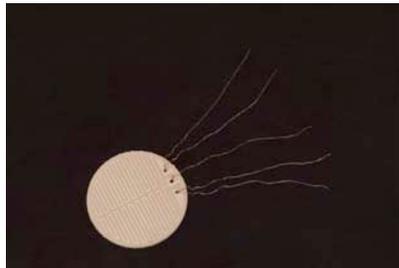
- Background
- Sensor System Development
- Sensor Testing
- Future Developments
- Summary



Background

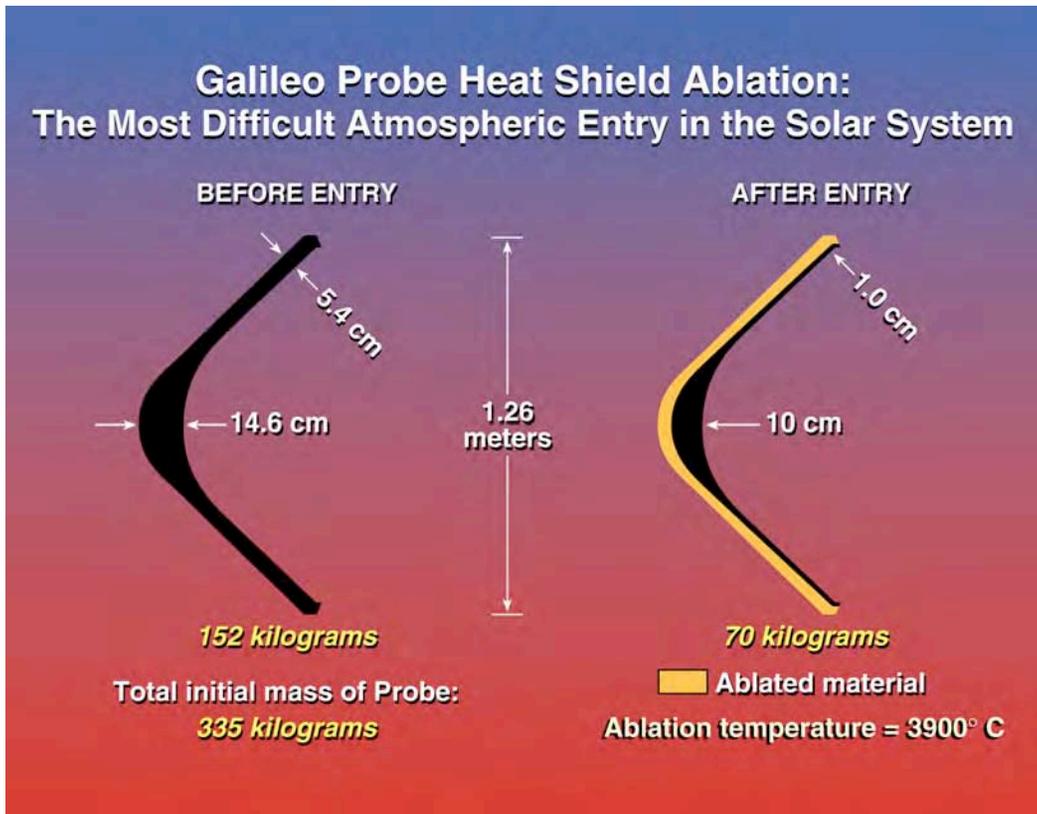


- Risk *reduction* requires risk *quantification*
- Risk quantification requires *entry physics sensors*
 - Flight Data + Math models + Ground test data
 - Aerothermal CFD, materials response models,
- **Sensors support engineering and science**





NASA Galileo Jupiter Probe Recession Sensor for Atmospheric Reconstruction



Analysis of Galileo Probe
Heatshield Ablation and Temperature Data,
Milos, et. al, Journal of Spacecraft & Rockets 1999

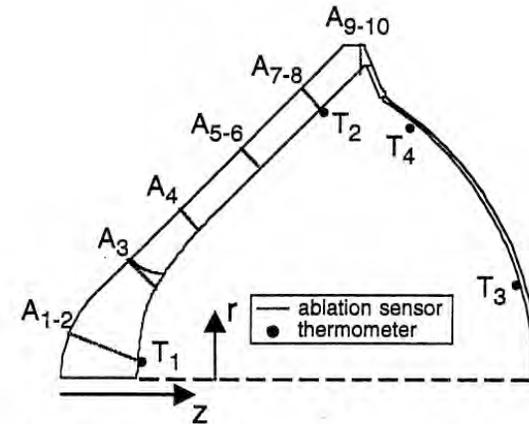


Fig. 2 Locations of 10 ablation sensors (A₁-A₁₀) in heatshield and four resistance thermometers (T₁-T₄) inside structure; sensors are not coplanar.

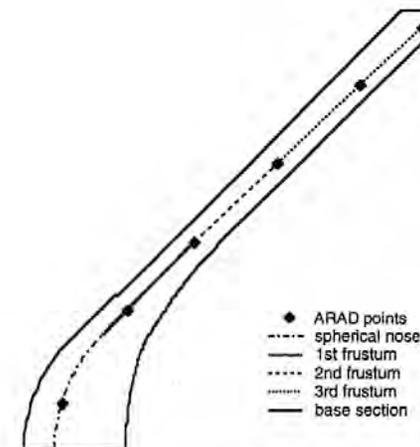


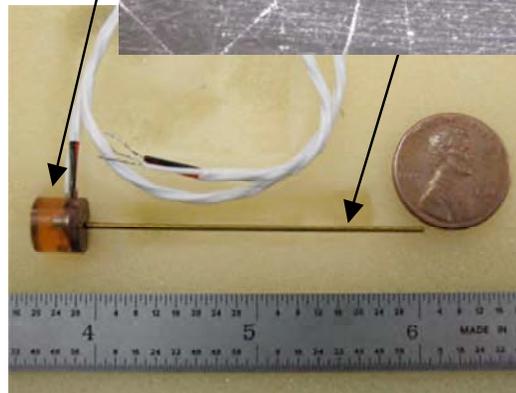
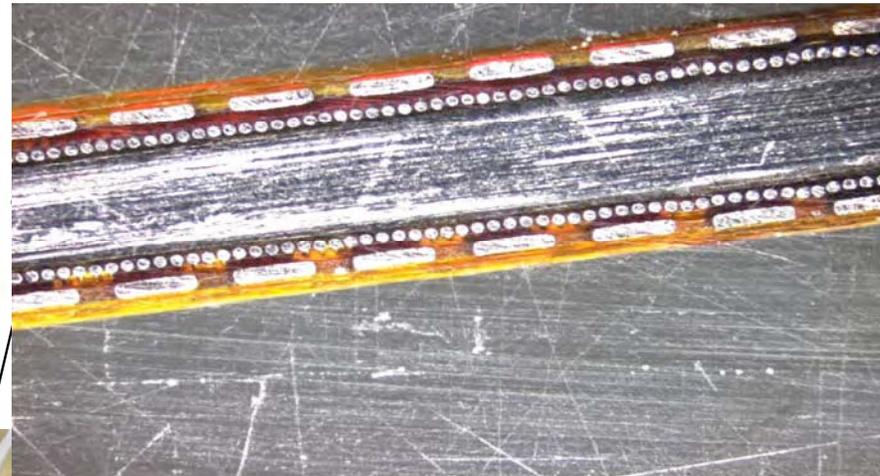
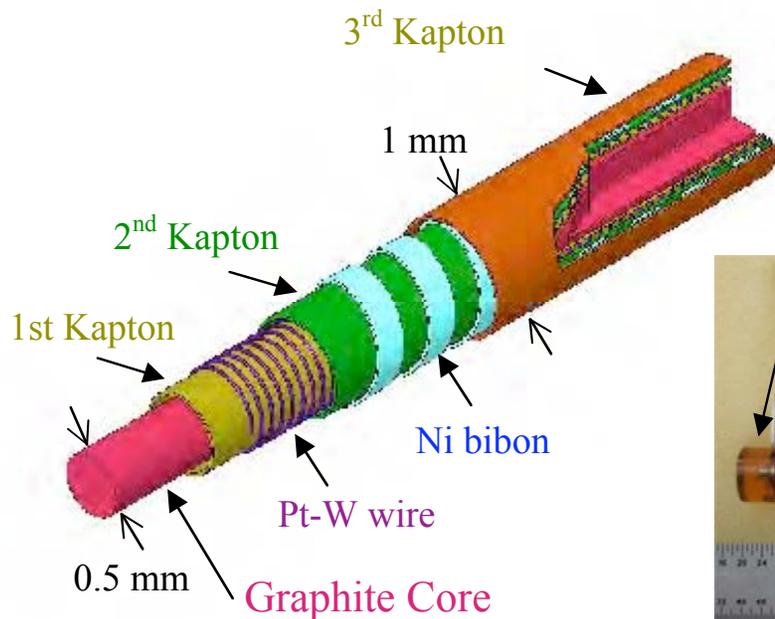
Fig. 5 Reconstruction of heatshield final shape (to scale with initial centerline thickness of 14.6 cm).



Heat Shield Recession Sensor ARAD Construction

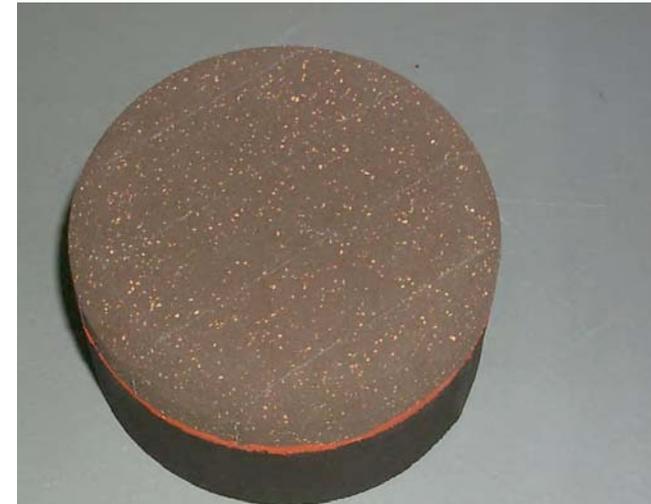
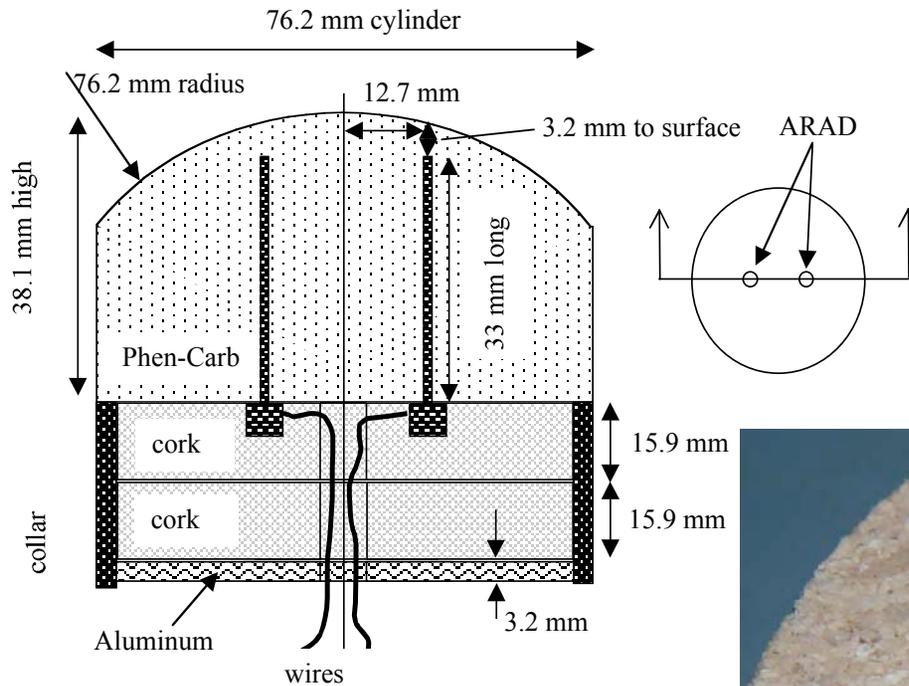


- Three coaxial conductive elements: Pt-W winding; Nickel ribbon; graphite core
- Kapton/epoxy provides a tenacious, electrically conductive char
- Measures a char zone - following a ~ 700 C isotherm
- Uncertainty of $\sim \pm 0.2$ mm - based on current source uncertainty of ~ 10 mV (0.91mm for Galileo)

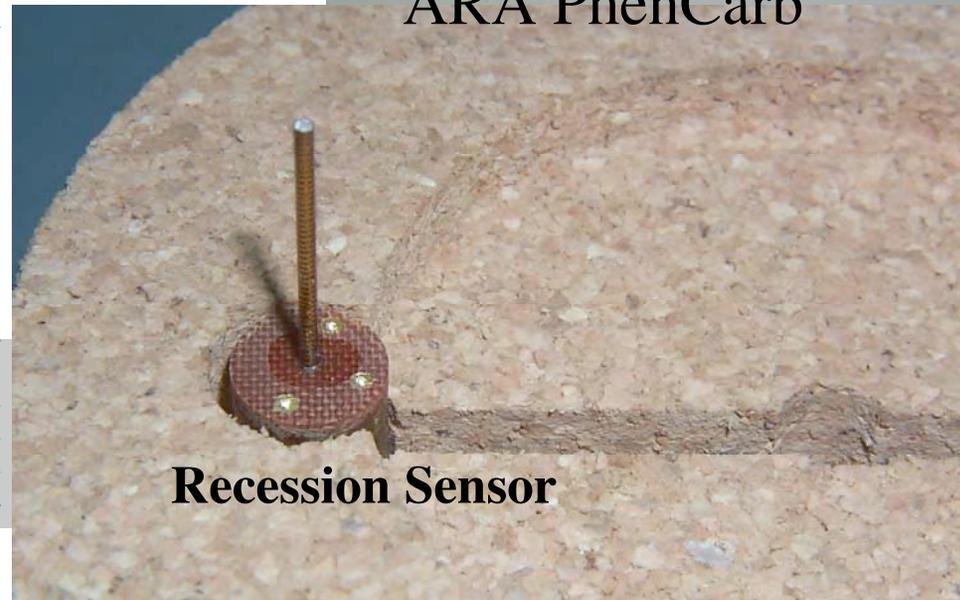




Integration of Models & Recession Sensors



ARA PhenCarb



Recession Sensor

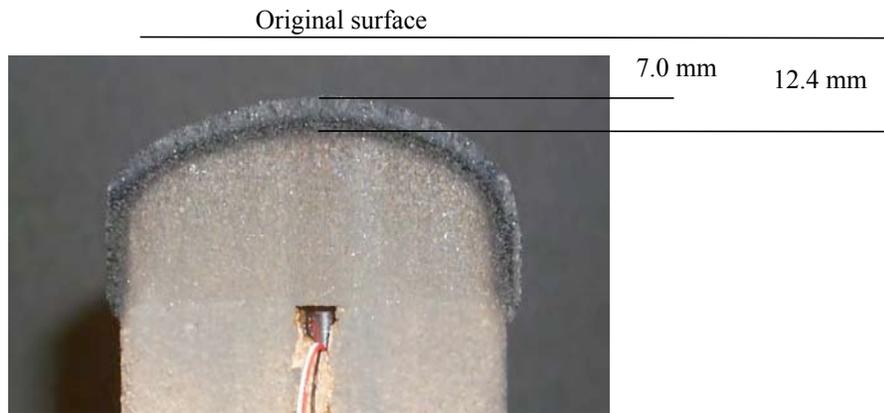
Model#	Density (lb/ft ³)	Density (g/cm ³)	Heat Flux (W/cm ²)	Duration (sec)
P24	24	0.385	610	145
P28	28	0.449	835	145
P32	32	0.513	1003	145



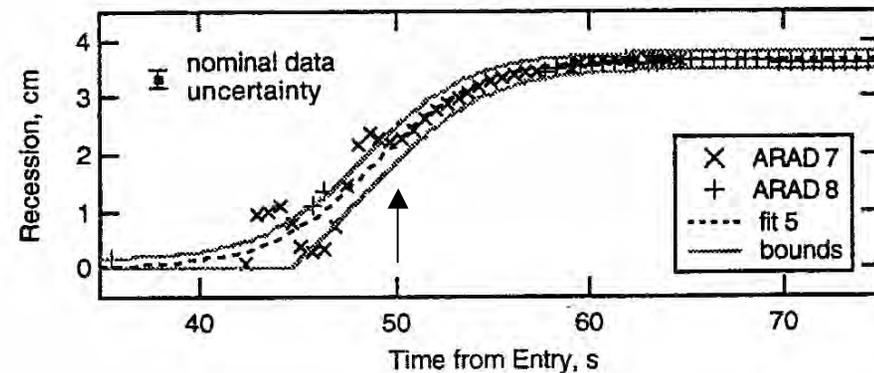
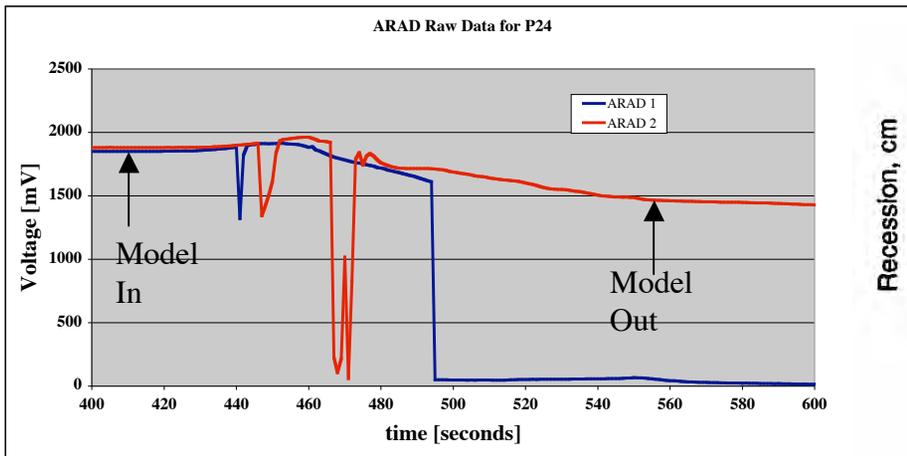
NASA Ames ArcJet Test Results



PhenCarb-28 Tested at 835 W/cm²



PhenCarb-24 Tested at 610 W/cm²



Resistance of ARAD for PhenCarb-24

30 June 2005

Analysis of Galileo Probe

Heatshield Ablation and Temperature Data,

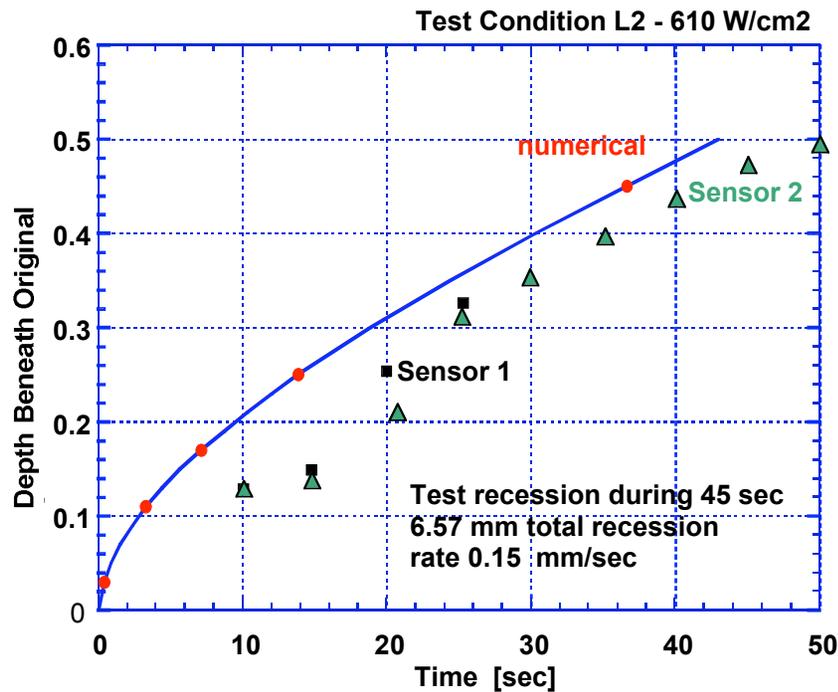
Milos, et. al, Journal of Spacecraft & Rockets 1999 *erm*



Sensor Response versus Material Model

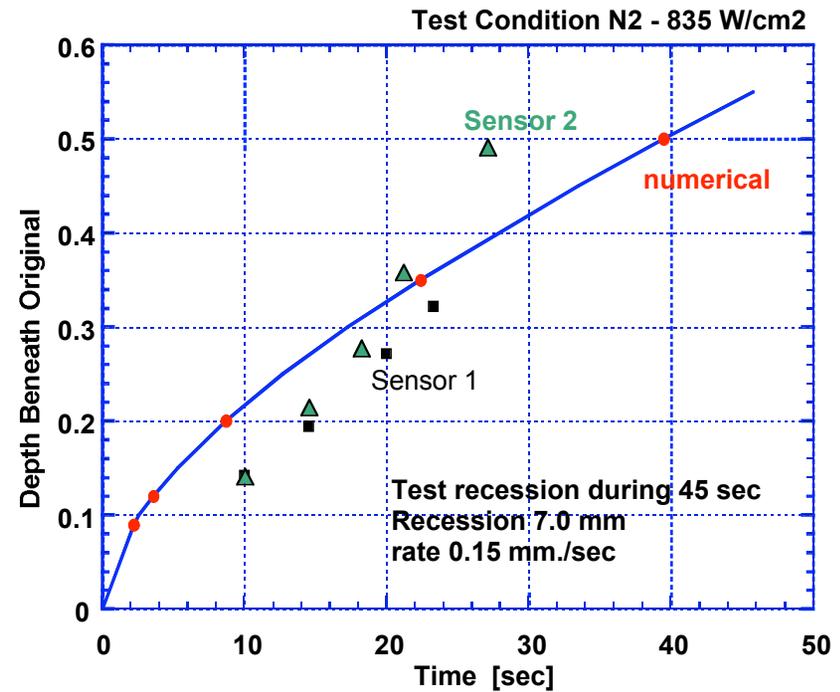


PhenCarb-24 Tested at 610 W/cm²



Movement of 700 Deg-C Isotherm for PhenCarb-24

PhenCarb-28 Tested at 835 W/cm²



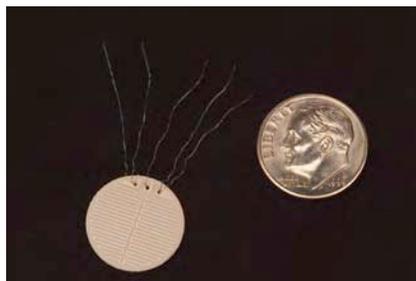
Movement of 700 Deg-C Isotherm for PhenCarb-28



Summary & Future Developments



- NASA has developed the basis for a **plug-and-play sensor system** for aeroshell traceability and science: Thermal, pressure, and **recession**.
- NASA has produced an updated recession sensor (**Ames ARAD**) for mid density ablators. They have been tested with two TPS materials for **traceability (leading to risk reduction)** of TPS and Aeroshell engineering performance suitable for Mars and outer planet missions.
- Potential application of thermal microsensors for extreme environments science, and **recession sensors for atmospheric reconstruction**.



30 June 2005

