

Exploration of Mars, Venus and Titan with Planetary Aerobots: A Pathway to Flight Readiness

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Workshop
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- Worlds with atmospheres and solid or liquid surfaces.
- Types of platform for solar system exploration
- In Situ Exploration with lighter-than-air vehicles
- Lighter than air vehicles for exploring
 - Mars
 - Venus
 - Titan
- Major Challenges
- Conclusions



Venus and Earth



Worlds with atmospheres and solid or liquid surfaces



**Venus Visible Image
Mariner 10**

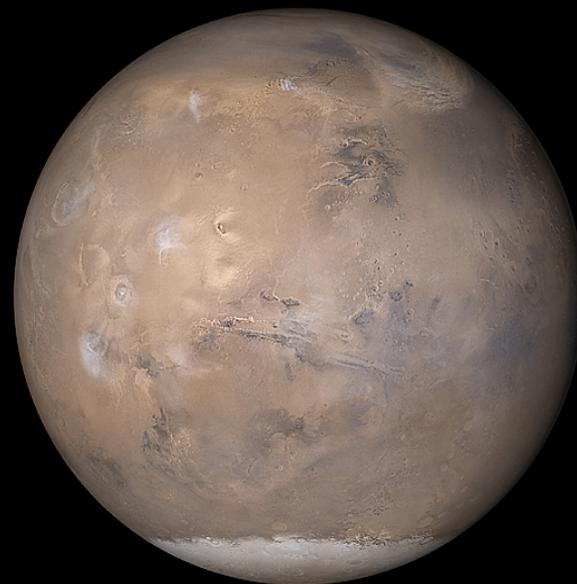
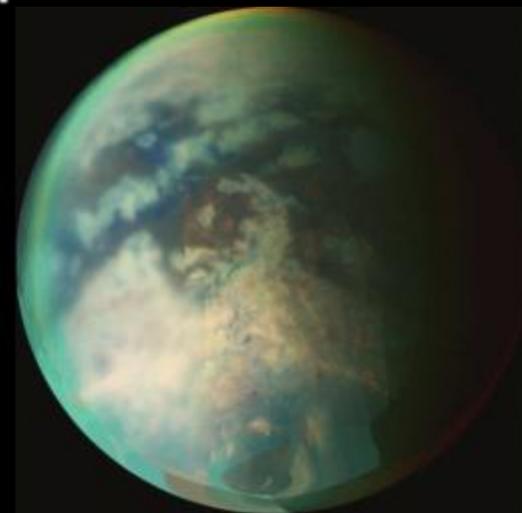


**Venus Radar Image
Mariner 10**





Earth, Titan and Mars: Worlds with atmospheres and solid or liquid surfaces





Atmospheric Environments of Venus, Titan and Mars?



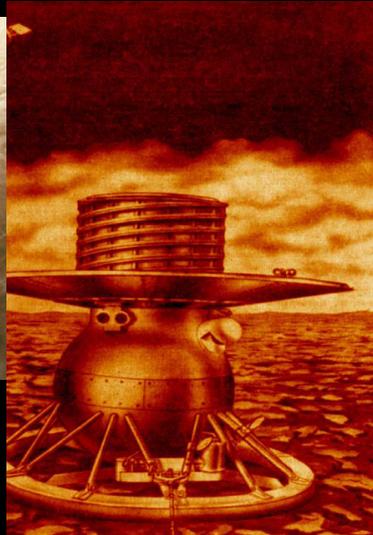
Parameter	Units	Earth	Venus	Titan	Mars
Temperature	K	290	750	90	155 - 300
Pressure	Bars	1	90	1.5	0.006
Density	kg/m ³	1.23	65	5.5	0.010



Flyby Missions
1962 – Present



Orbiter Missions
1971 – Present



Descent Probe & Lander Missions
1972 – Present



Rover Missions
2004 – Present

**Spacecraft
Platforms**

**In Situ Exploration
Platforms**

**Surface
Mobile**

**Fixed
Site**

**Aerial
Mobile**



Concept for Mars Superpressure Balloon - 1998



Entry Interface
 $t = 0$ s
 $h = 125$ km

Parachute Deploy
 $t = 160$ s
 $h = 19$ km

Heat Shield
Jettison
 $t = 170$ s
 $h = 18.5$ km

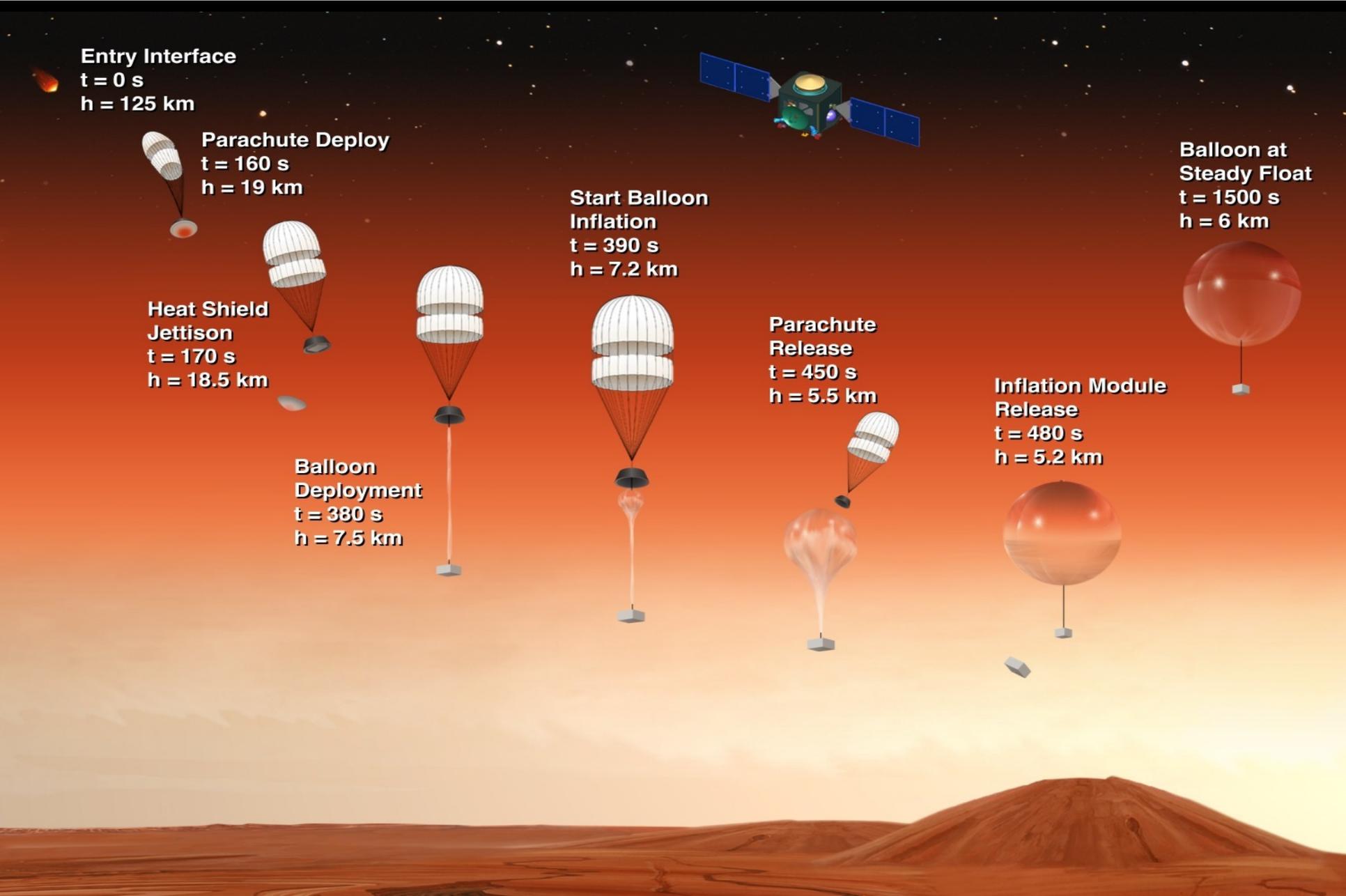
Balloon
Deployment
 $t = 380$ s
 $h = 7.5$ km

Start Balloon
Inflation
 $t = 390$ s
 $h = 7.2$ km

Parachute
Release
 $t = 450$ s
 $h = 5.5$ km

Inflation Module
Release
 $t = 480$ s
 $h = 5.2$ km

Balloon at
Steady Float
 $t = 1500$ s
 $h = 6$ km



Pumpkin Balloon



- Polyethylene film and PBO tendons
- Tendon reinforcement along each gore
- Tendons carry most of the superpressure load allowing for thinner balloon films
- Tendons carry most of the deployment shock loads

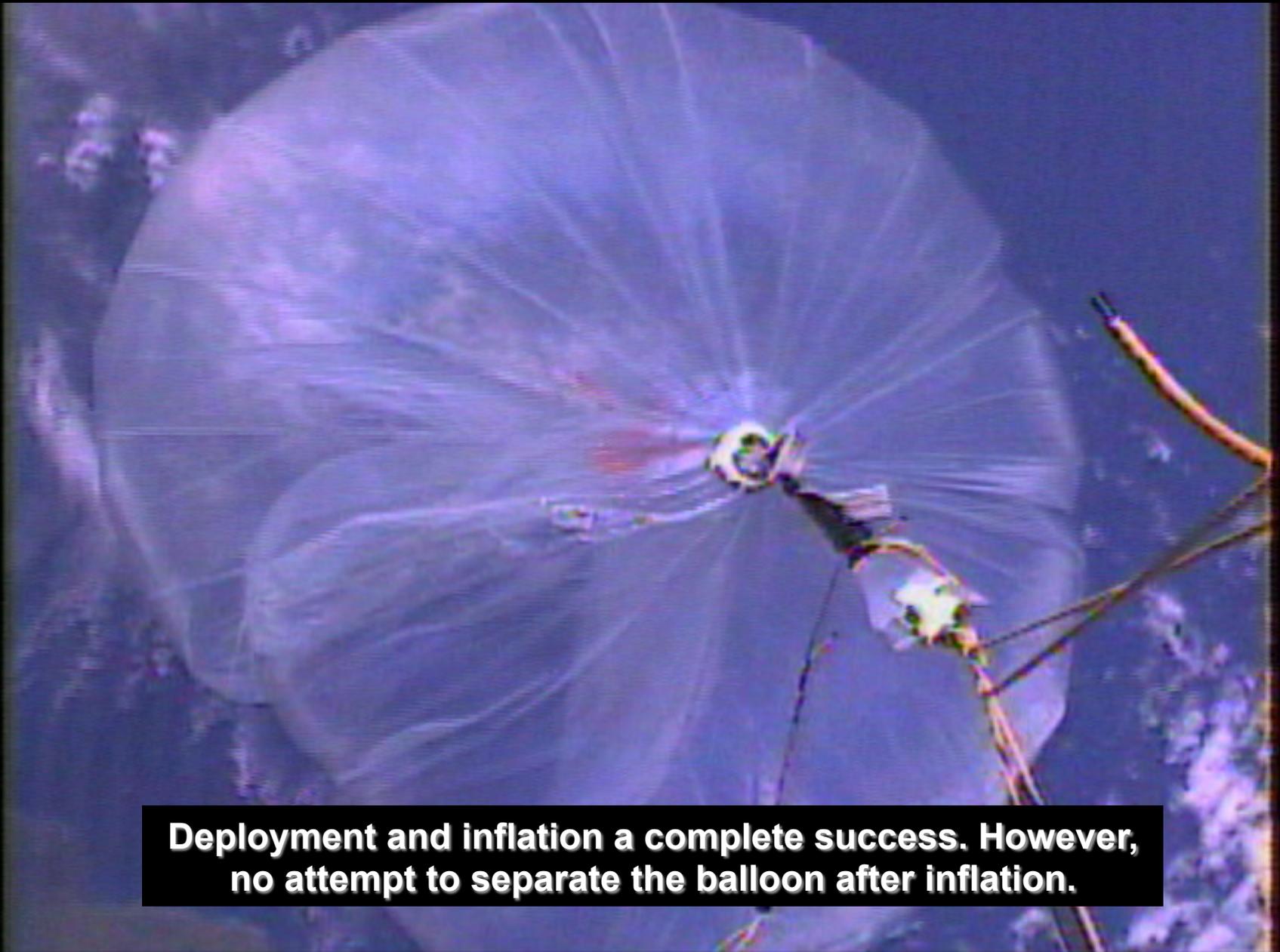
Spherical Balloon



- Mylar film with taped seams
- Extra thick end caps and
- No other reinforcement



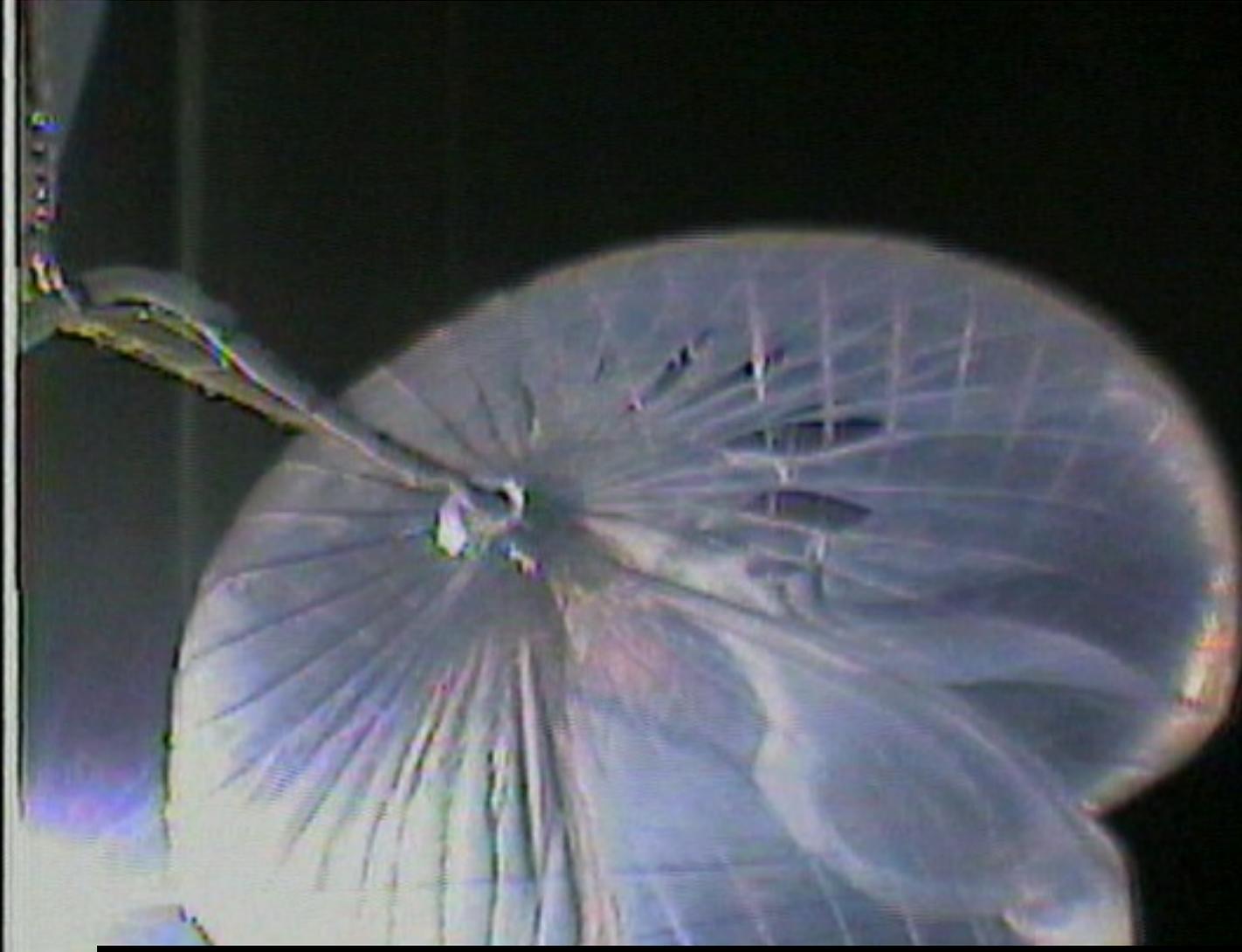
Prototype Mars Superpressure Balloon Deployment Test in Stratosphere (2006)- Spherical Balloon



Deployment and inflation a complete success. However, no attempt to separate the balloon after inflation.



Prototype Mars Superpressure Balloon Deployment Test in Stratosphere (2006)- Pumpkin Balloon

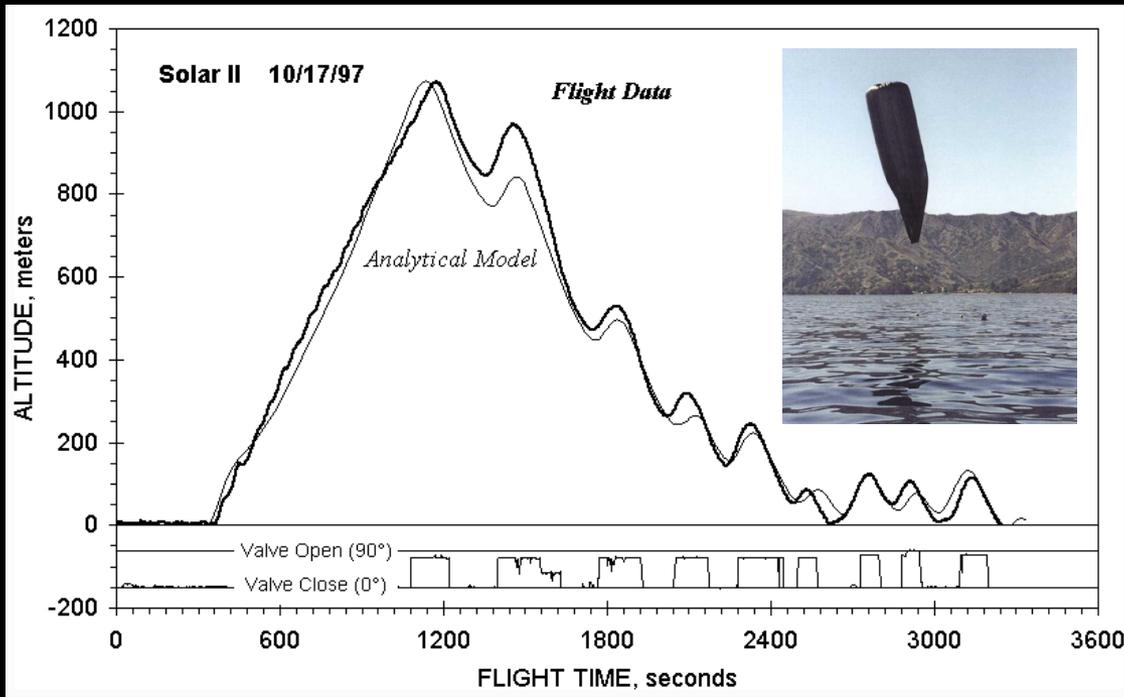


Parachute failed to fully deploy causing dynamic pressure on the balloon during inflation to be more than double the planned value (12 Pa vs 6.5 Pa).

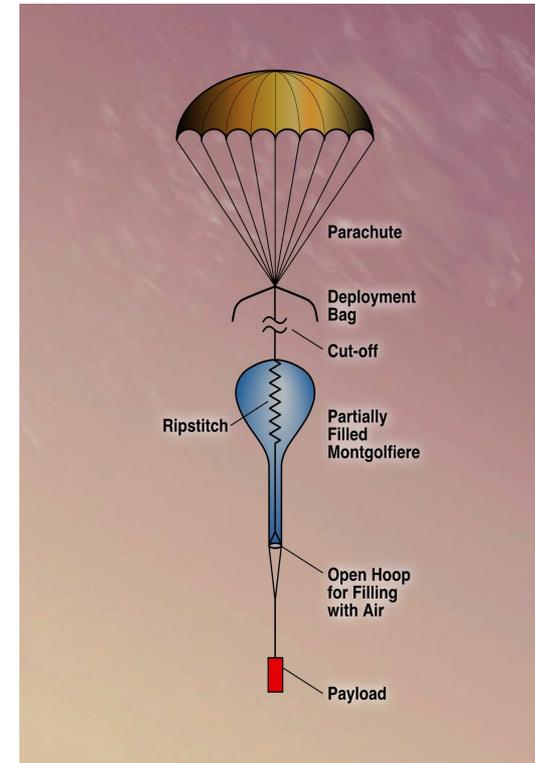
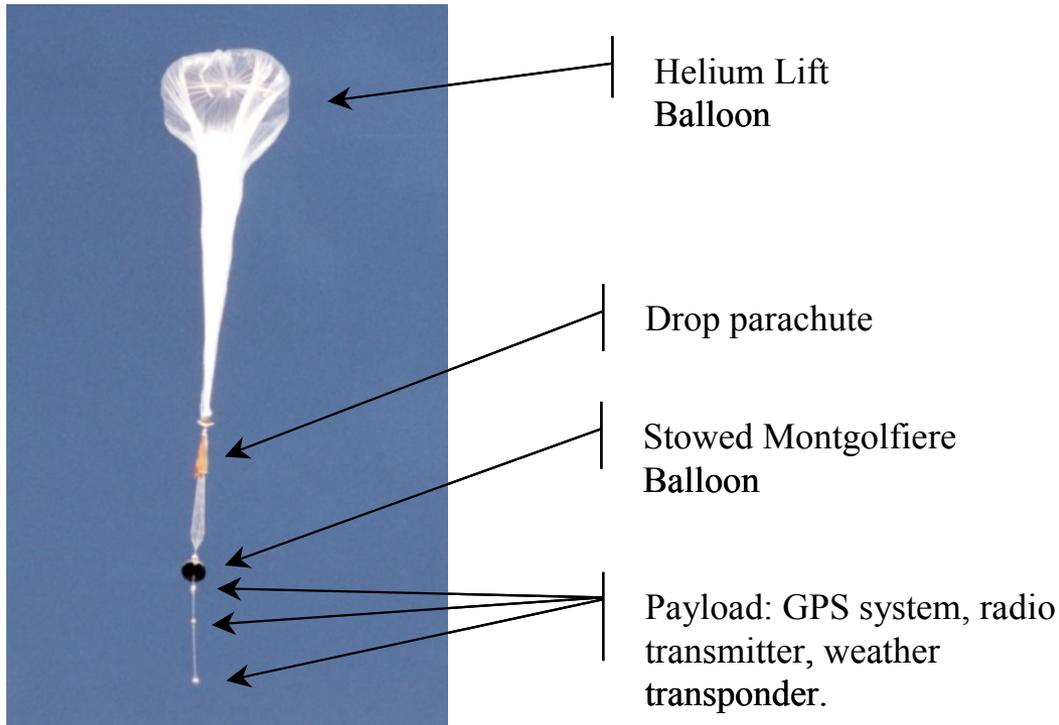


- An end-to-end demonstration of deployment, inflation and float in the stratosphere is still needed.
- Two more flight tests of the superpressure balloon are planned for June 27 and June 29 2007
 - A 12 m diameter sphere (900 m³, 70% larger volume than 2006) – providing 5kg payload capability at Mars.
 - A 660 m³ pumpkin (15% larger volume than 2006)
 - Both tests will include parachute and inflation system separation and subsequent balloon float
- No further superpressure balloon tests are currently funded

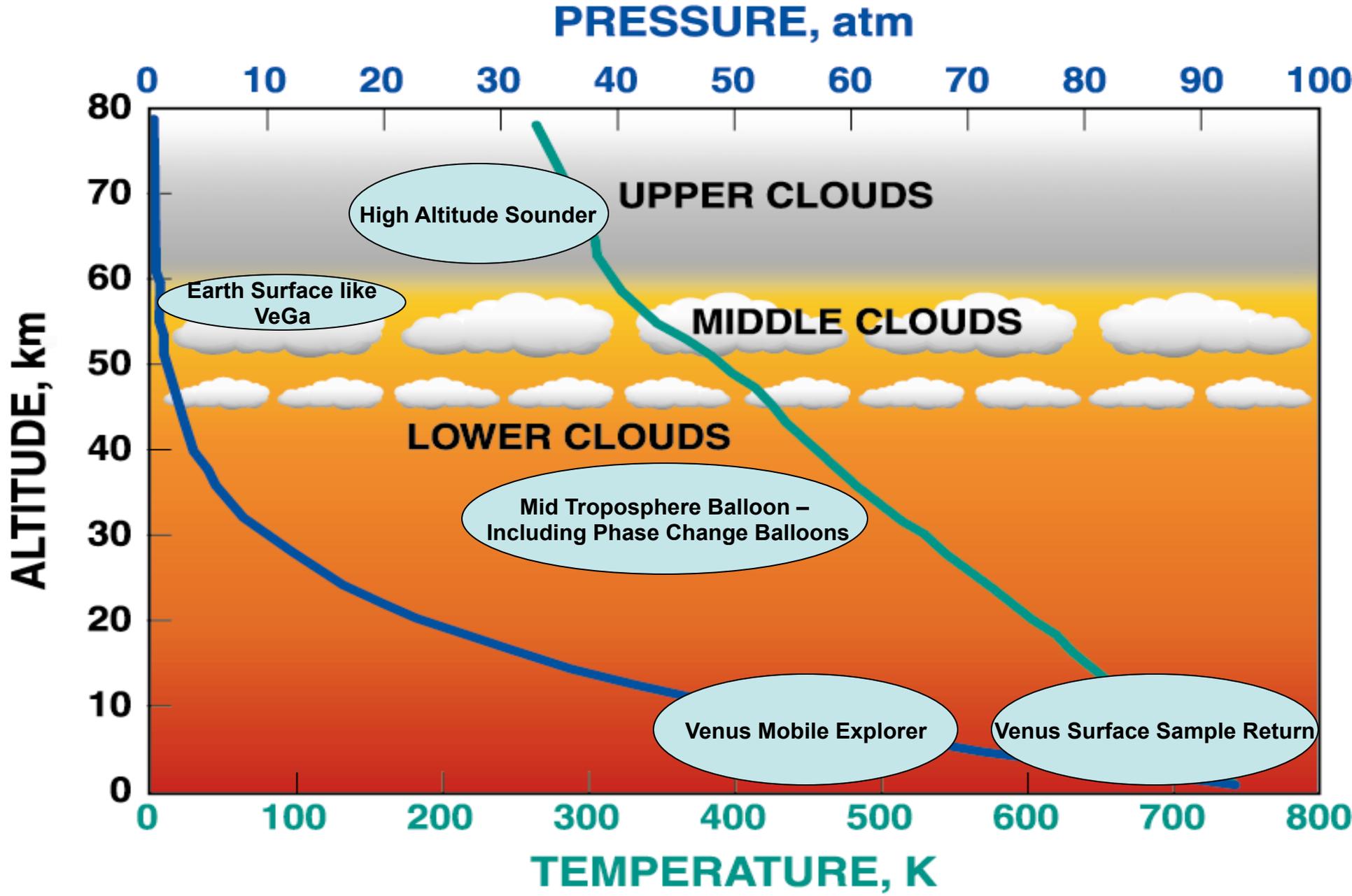
If successful, the June stratospheric deployment tests will retire most of the risks associated with sustained flight of payloads of a few kgs for focused science objectives on Mars



These tropospheric tests have established the principle of altitude control with a vent at the top of the balloon.



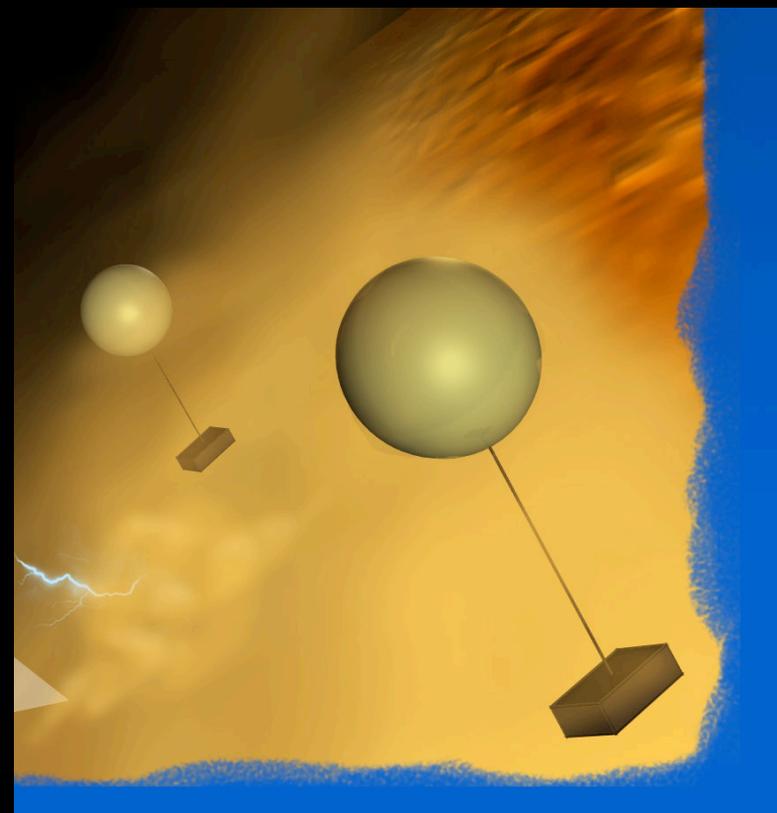
- Several Earth stratospheric deployments have been conducted.
- Several tests with a 15-m diameter balloon have been successful
- **A 30-m test is pending in July, 2007.**





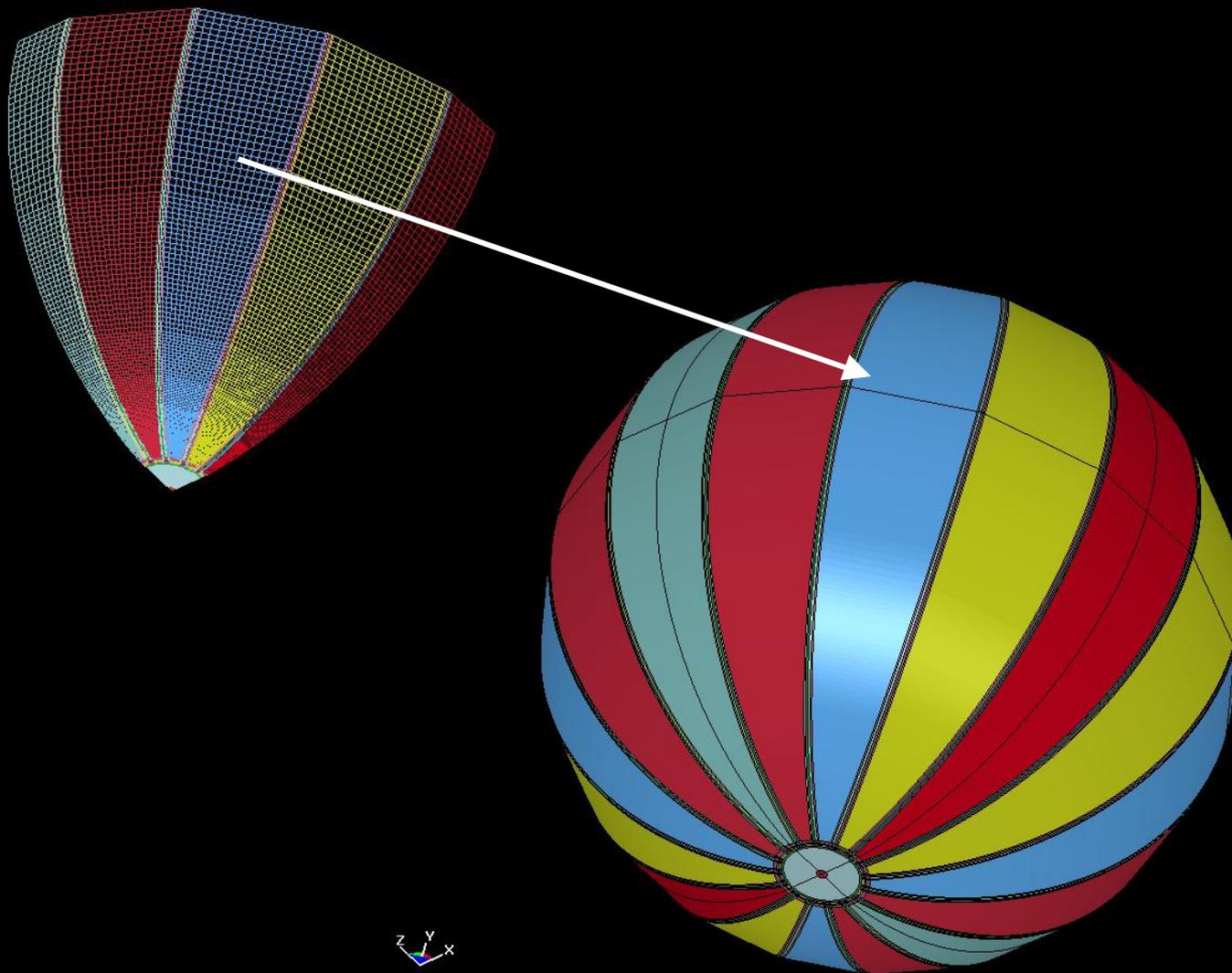
Venus High Altitude Balloons VEGA Mission, 1985 – Earth Test Flight







Venus High Altitude Balloon - Finite Element Modeling

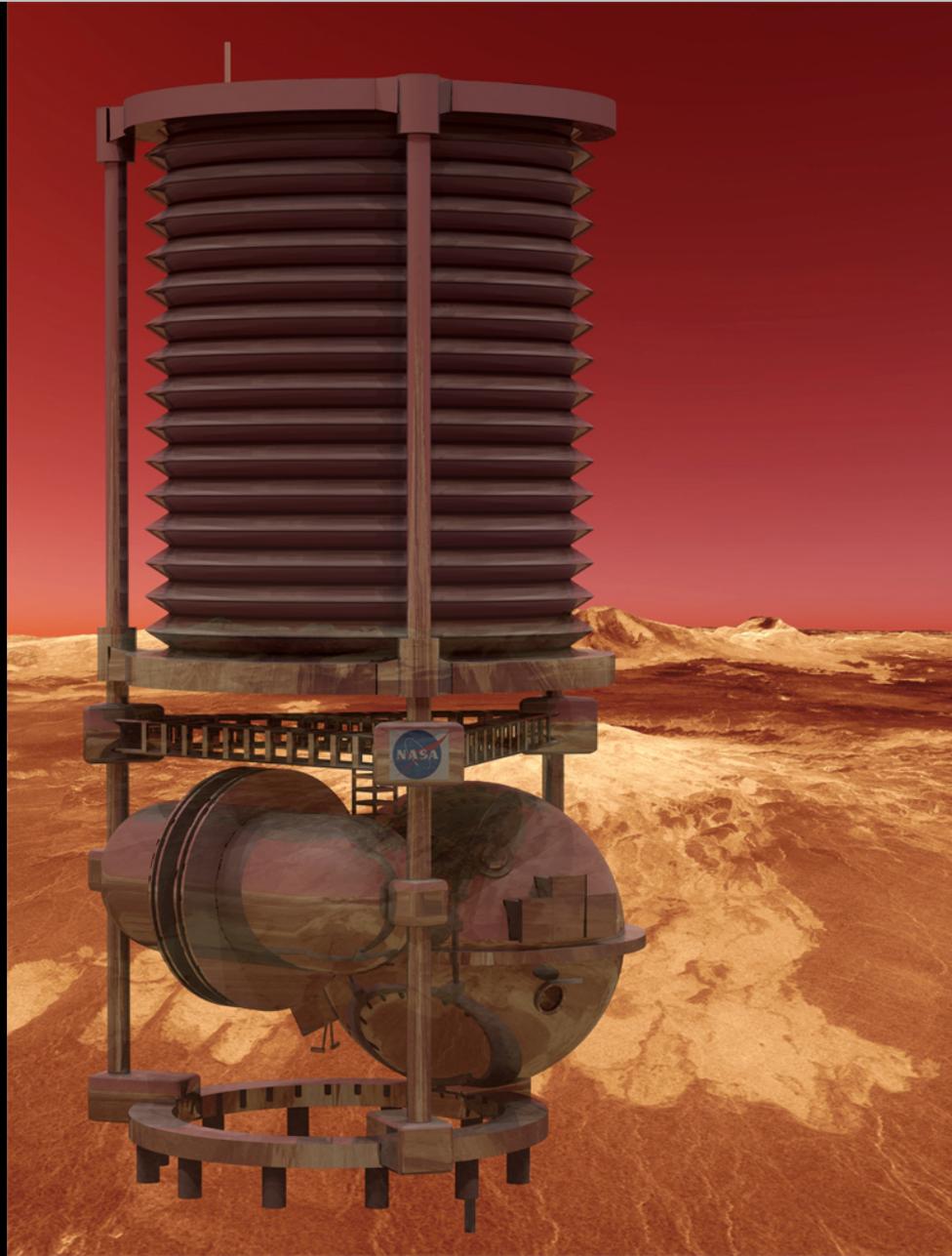


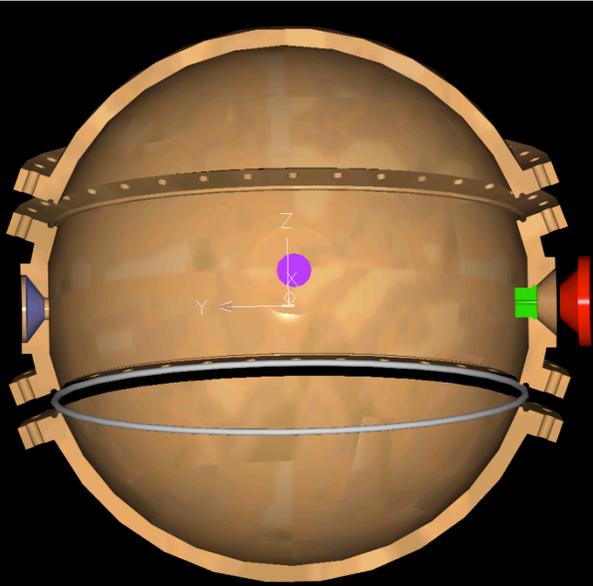


- Target altitude is ~20km where temperature is ~250C, pressure 15 bars.
 - Measurements there may be key to unraveling the mystery of Venus superrotation.
- High temperature electronics for <300C is available
 - CMOS compatible Silicon on Insulator (SOI) MESFETS (see paper by [Ervin et al, IPPW4](#))
- Balloon materials demonstrated at 300K with adequate strength for superpressure balloon [Yavrouian et al, 1999](#)
 - Kapton FN – coextruded combination of Kapton and Teflon
 - Polybenzoxazole film is an alternative
- Proof of principle of **phase change balloons** demonstrated at JPL in the ALICE program in 1990s with refrigerants. Need to validate with water or ammonia balloons at Venus temperatures

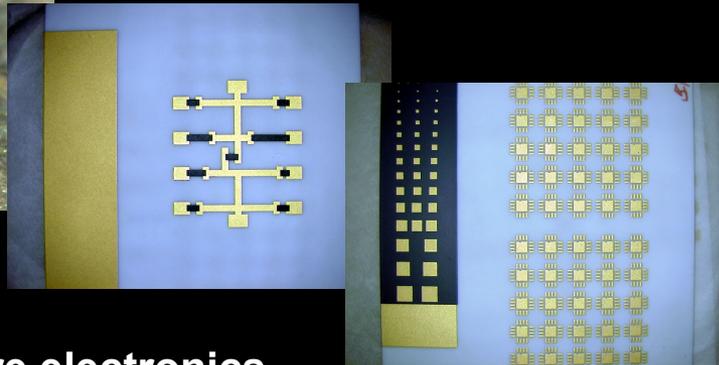
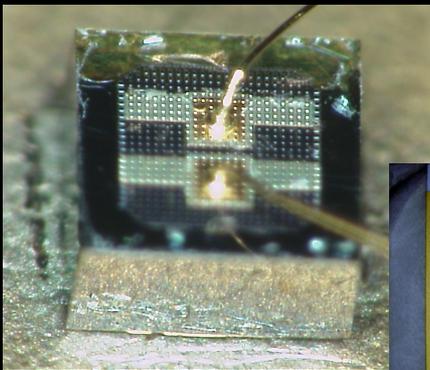


Venus Mobile Explorer – Exploring the Surface of Venus





Pressure vessel & insulation



High temperature electronics

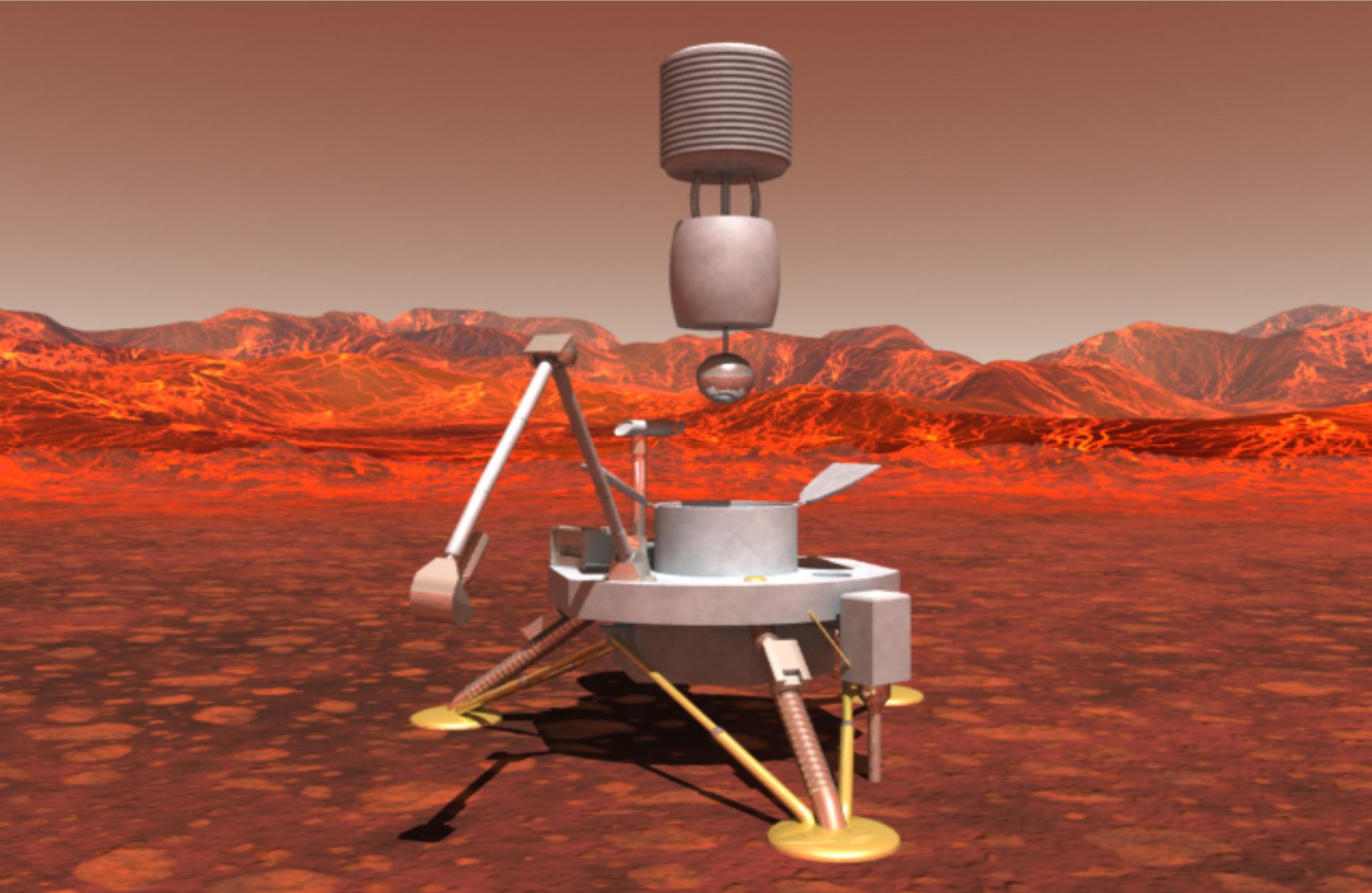


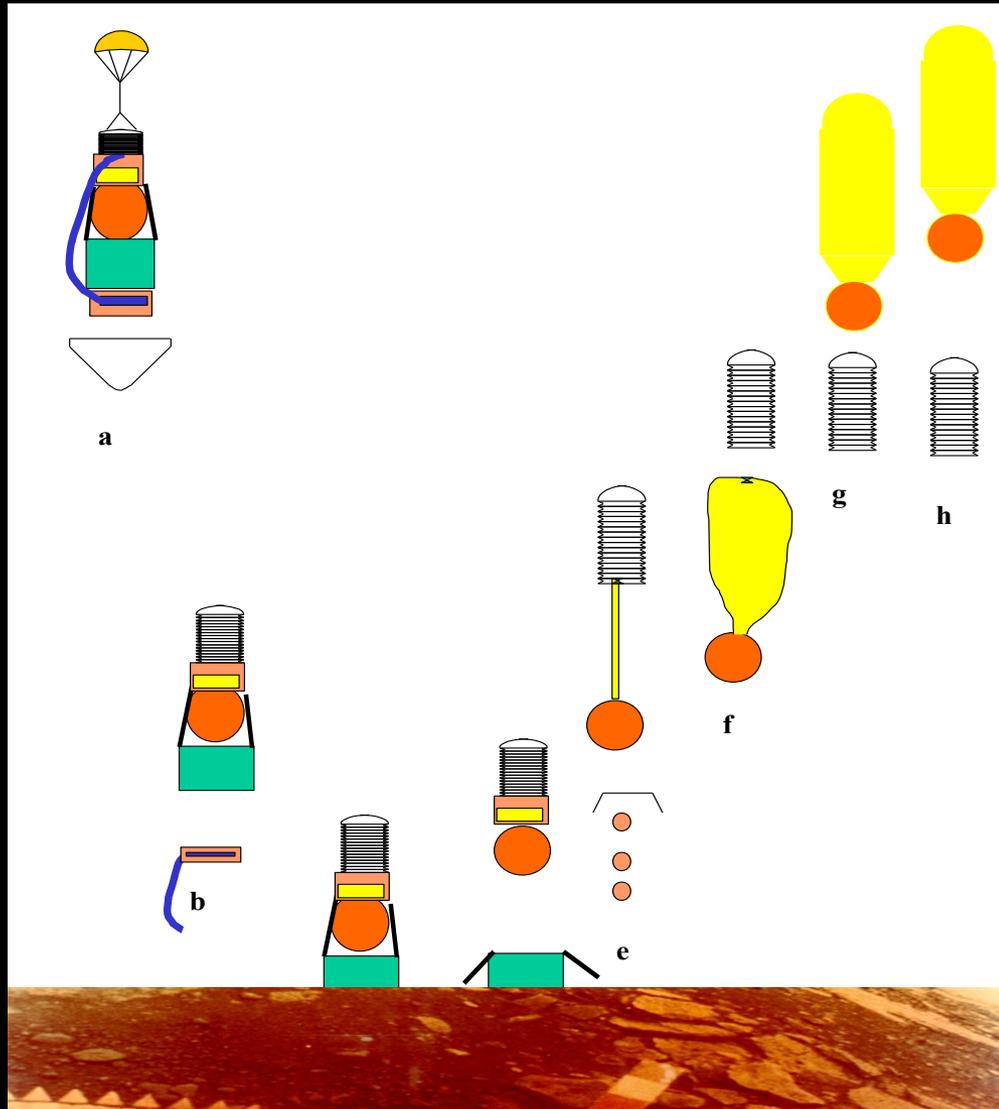
Metal bellows buoyancy device



Venus Surface Sample Return

Bringing a sample from the Venus Surface back to Earth

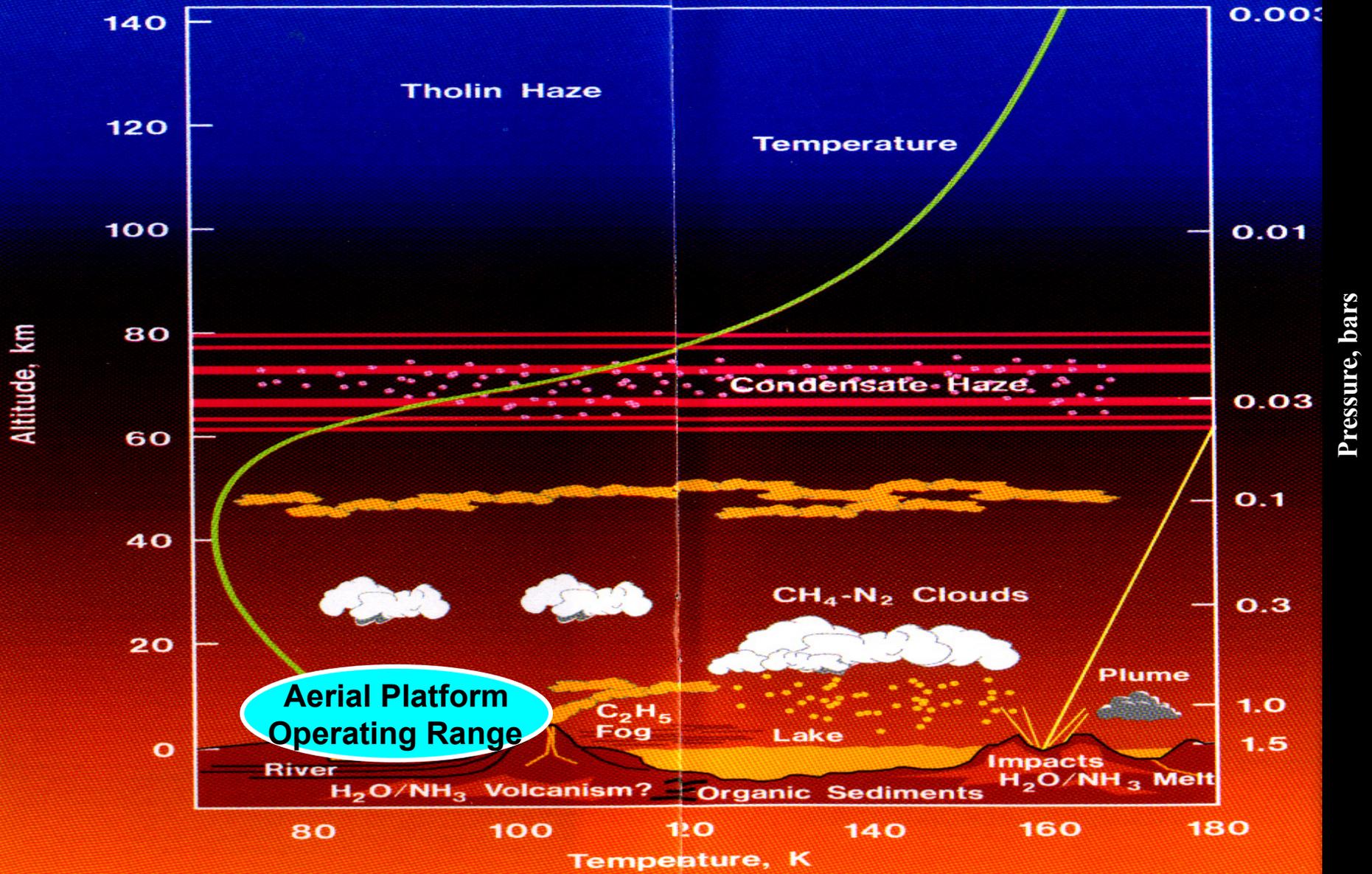


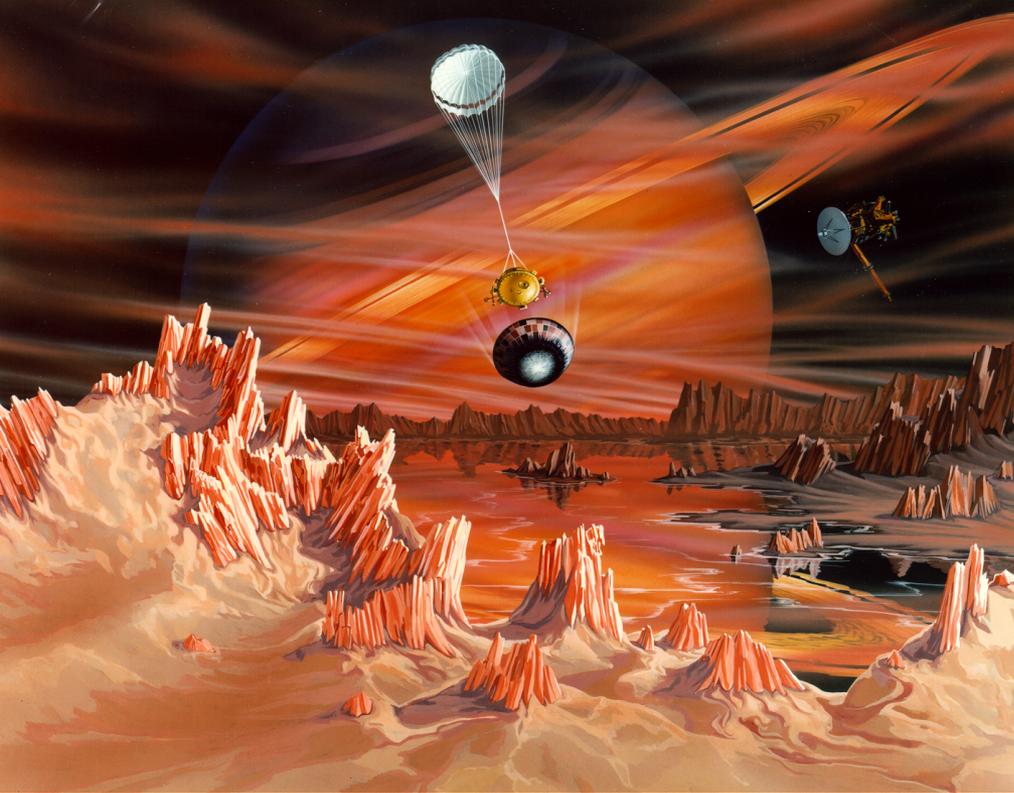


Venus Dual Balloon concept, Kerzhanovich et al IPPW4



- **Next step** is a superpressure balloon flying in the same regime as the VeGa balloons
 - Large and more sophisticated payload than VEGA
 - Longer duration flight with multiple circumnavigation
 - Technology is ready
- Technologies for the extreme environments of Venus – discussed at IPPW4 – are critical to all further scenarios and there are many interrelationships and **feed forward opportunities**.
- Next step will depend on not only technology readiness but the value of the science that the mission can yield.
- Venus Mobile Explorer which can open up the surface of Venus to detailed in situ exploration may be a major driver for future development.
- Venus Surface Sample Return is impossible without a balloon capability that can provide transit from the surface to ~65km

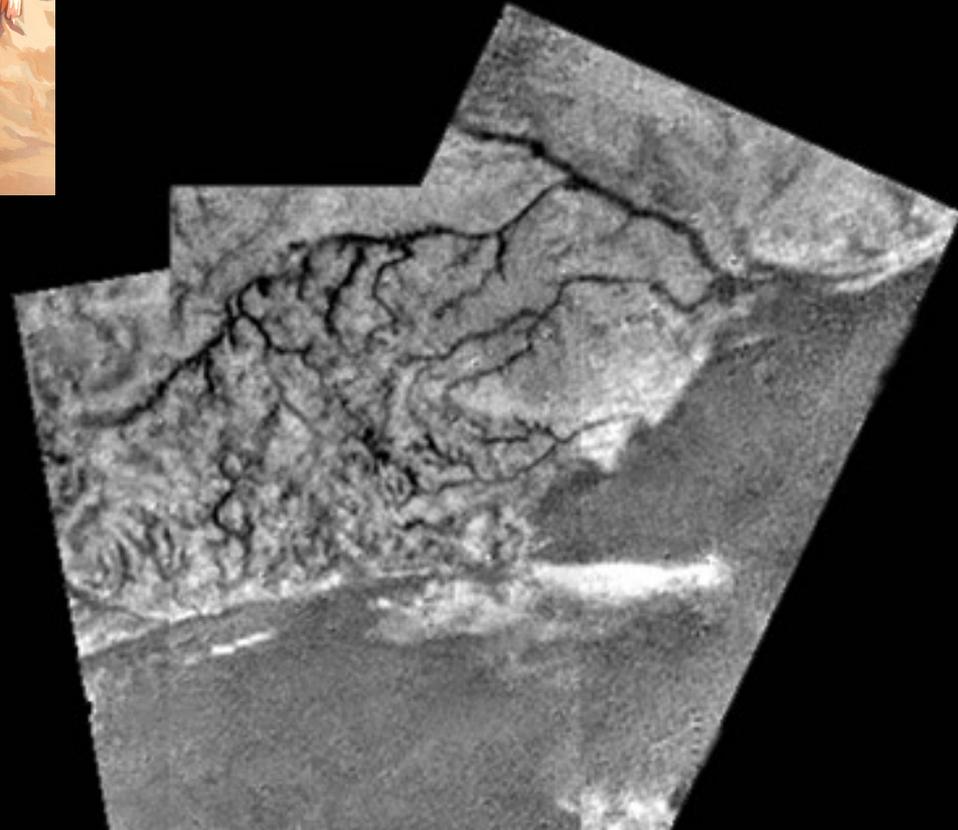




Aerial imaging from descending Huygens probe reveals that in the lower atmosphere:

- **It is cloud free**
- **Winds are light**

Spectacular landforms are seen from the descending probe



Montgolfiere (Hot air) Balloon

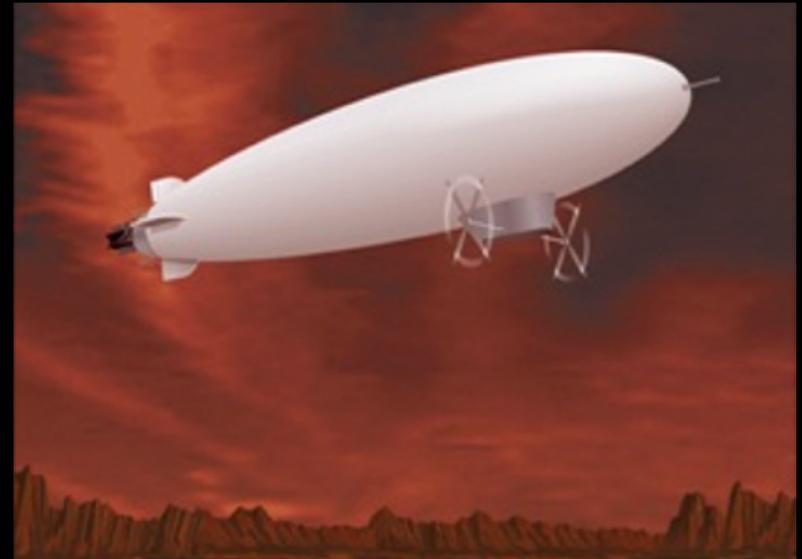
Wind-driven

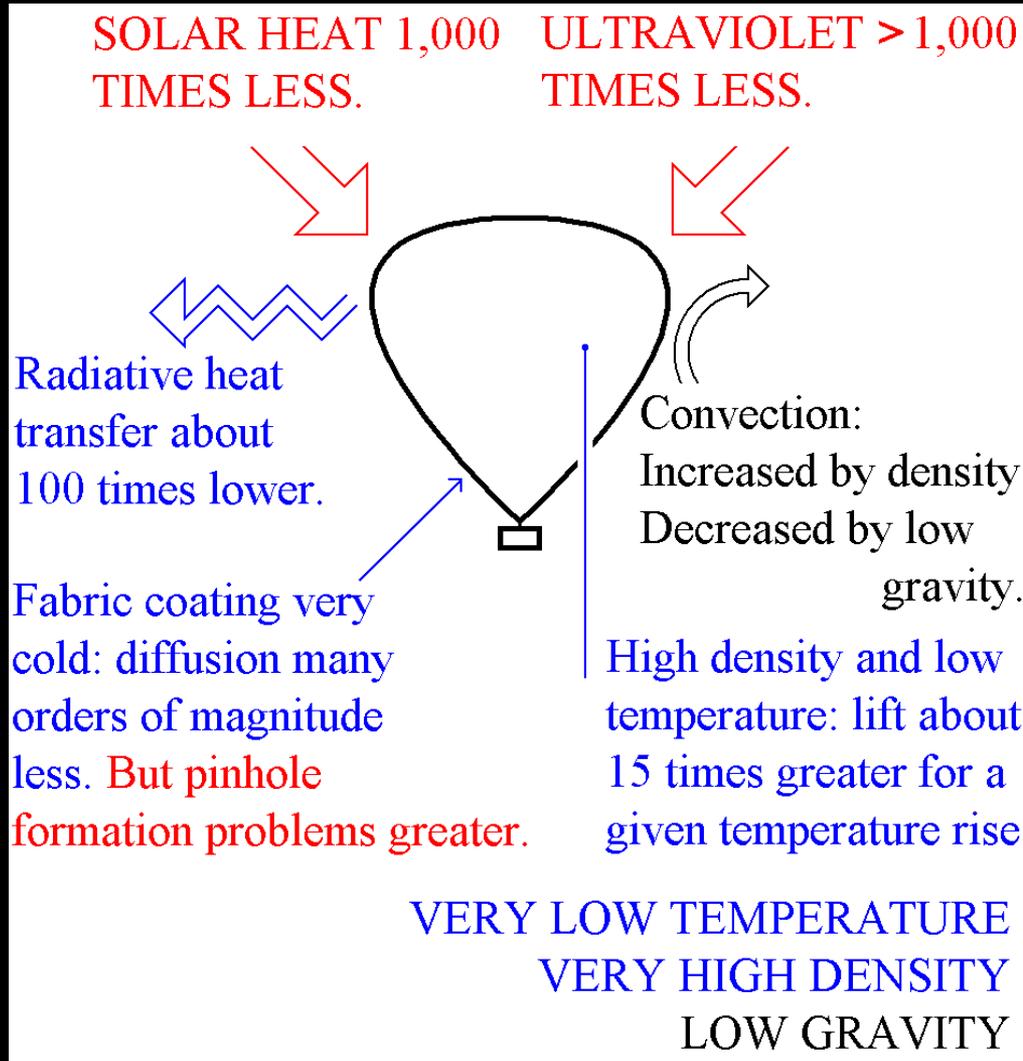


Self-propelled



Self-propelled Blimp





Julian Nott, Titan the Ultimate Destination for Aerobots, IPPW4 Proceedings, 2006



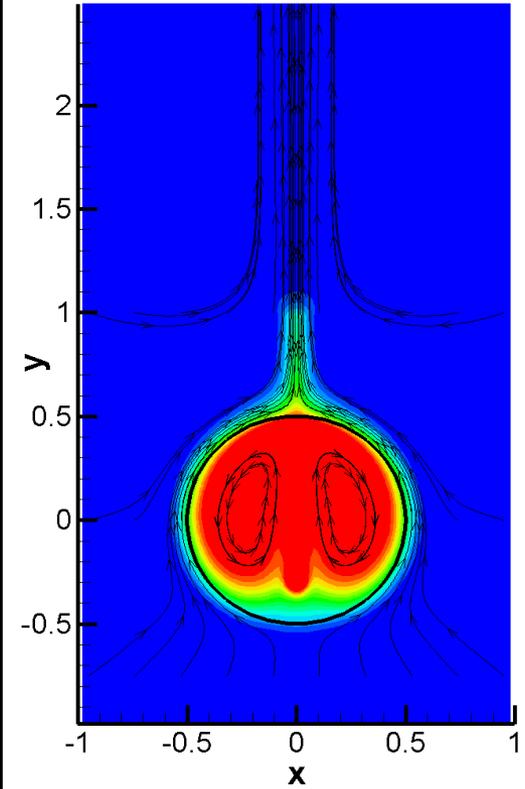
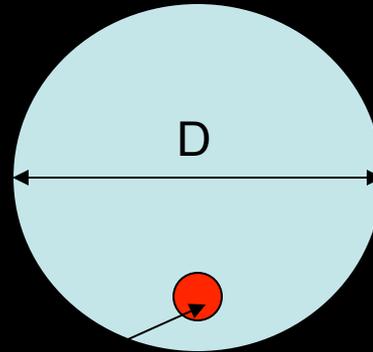
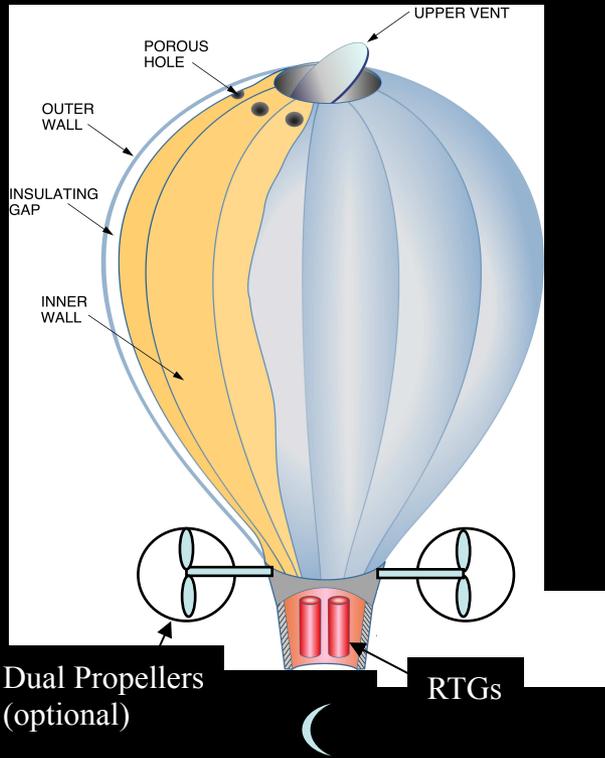
**Gelbo Flex Test of
Polyester composite film
2000 cycles at 90K**



Blimp Hull at Cryogenic Temperatures.

Hall et al, Proceedings of COSPAR, 2004

Frame 001 | 03 Apr 2007 | IBFS V1.0 AT ITIME= 40000





Aerobot Aerodynamic Model/Simulator



AS800B Airship Simulation



Airship Simulation Control

Time Control

Run	0.40	0.05	0.02	1	Step
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Dgui Ssgui Close Gui Quit

SIM Dspace airship

Airship Info

Type:	AS800B
Length:	11.43 meters
Diameter:	2.44 meters

Reset Position

Airship Location

X (m):	0.00	Vx (m/s):	0.06
Y (m):	0.00	Vy (m/s):	0.00
Z (m):	15.00	Vz (m/s):	0.02
Roll (deg):	-0.0	RollRate (deg/s):	-0.2
Pitch (deg):	-4.0	PitchRate (deg/s):	0.4
Yaw (deg):	0.0	YawRate (deg/s):	0.0

Controls

Thrusters		Effective Rudder	Effective Elevator
Angle (deg)	Thrust (N)	Angle (deg)	Angle (deg)
25.0	17.6	-35.0	0.0
Zero		Zero	Zero

Dspace Mass/Buoyancy

Show Vehicle Axes

Forces

Show Fin Aeroforces

Show Stabilizer Aeroforces

Paper by Elfes et al at this meeting IPPW 5





- Titan is in many respects the ideal environment for surface exploration with a lighter-than-air vehicle
 - Dense and cold high molecular weight atmosphere
 - Little diurnal temperature variation and almost no UV radiation
 - Clear skies beneath the clouds and light winds at the surface
- A Titan aerobot can acquire:
 - Imaging and remote sensing surveys over a global traverse at resolutions ranging from meters to centimeters.
 - In situ sampling from a great diversity of terrains ranging from organic dunes, to cryovolcanic features to methane lakes.
- Although the principles of Titan aerobot operation are established, the practice is a long way from validation:
 - Significant investment in the aerial vehicle technology, instruments and cryo sampling methods
 - End-to-end simulations of aerobot deployment & operation & where feasible actual validation in cryogenic environments.

- Mars
 - Scientifically, aerobots occupy a **niche role** complementing the science that can be done with orbiters and surface rovers.
 - Superpressure balloons technology, capable of carrying payloads of a ~5 kg for mission durations of months is now within reach.
 - Solar Montgolfiere balloons with larger payloads but capable of long duration flight only in the summer polar regions may soon be demonstrated.
- Venus .
 - Scientifically, lighter-than-air technology must play a **central role** in the future in situ exploration and surface sample return from Venus.
 - Superpressure balloon technology for payloads of tens of kgs. over flight times of weeks to months at altitudes of 54-56 km is at hand.
 - Metallic bellows balloon technology can enable long duration mobile exploration of the Venus surface and ultimately Venus surface sample return.
- Titan:
 - Scientifically, a Titan aerobot provides the ability to carry out both **remote sensing** from beneath the Titan clouds and haze and **mobile in situ sampling** for a diverse range of terrains from organic dunes to methane lakes.
 - **Radioisotope Montgolfiere technology** offers the potential for long duration (years) operations for altitudes from 0 to 10 km with controlled surface sampling.
 - While the principles of Titan aerobot operation are established, the technology validation **sufficient for commitment to a Flagship mission** is still needed.



- ILC Dover – Venus mid altitude balloon
- Lamart Corporation – cryogenic balloon material for Titan
- Lyntech Corporation - Hydrogen Recovery System
- Winzen - Titan Montgolfiere design
- LSTC – Venus Balloon Finite Elements modeling
- GSFC/Wallops - Balloon design, lab testing and fabrication oversight
- Near Space – Titan balloons, Mars balloons, stratospheric test
- Tensys Corporation – Titan airship design
- Old Dominion College of Engineering & Technology – modeling simulation and visualization
- Johns Hopkins Applied Physics Lab – Titan Flagship Mission Study
- Jet Propulsion Laboratory – Viktor Kerzhanovich, Jack Jones, Kim Reh, Tibor Balint and others for help with this presentation.

Special Thanks: To Jacques Blamont whose vision of balloon exploration of the solar system played a key role in the establishment of the planetary aerobot initiative at JPL.